

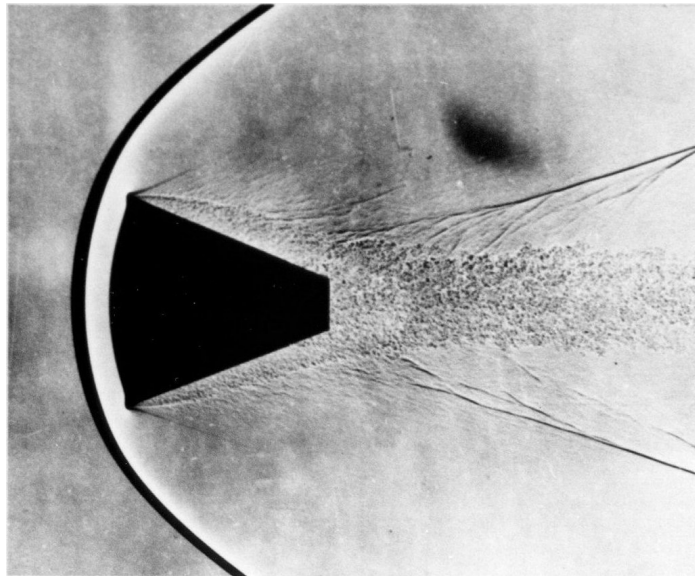
ESCOAMENTO COMPRESSÍVEL

EXERCÍCIO

PROPRIEDADES APÓS CHOQUE NORMAL



Uma onda de choque normal ocorre na seção de testes de um túnel de ventos supersônico. A montante da onda $M_1=3$, $P_1=0,5\text{atm}$ e $T_1=200\text{K}$. Encontre Mach, pressão, temperatura e massa específica após o choque.



<<https://commons.wikimedia.org/wiki/File:Heatshield-test3.jpg>(27/04/2020)>

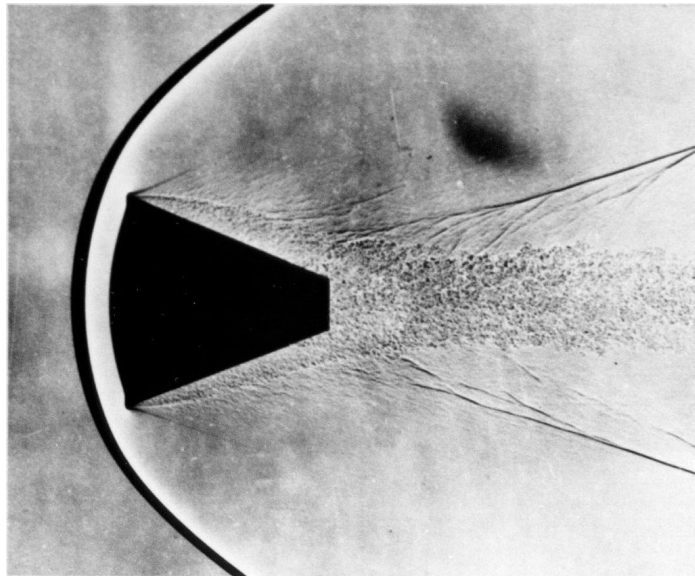
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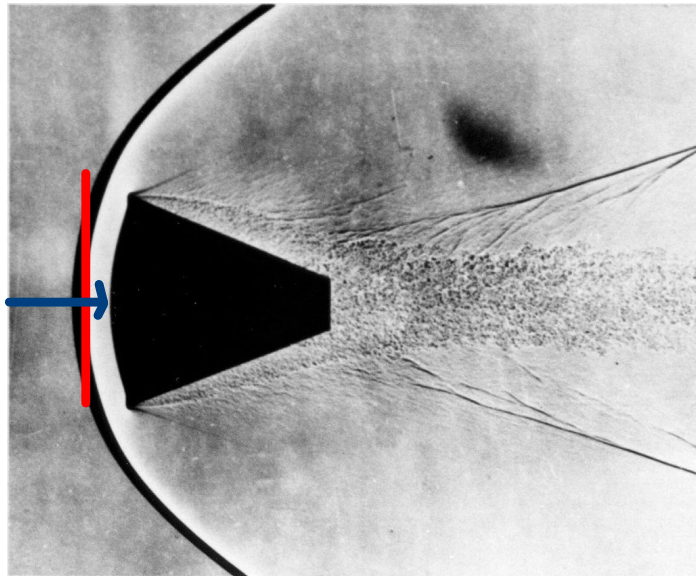
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ESCOAMENTO COMPRESSÍVEL

EXERCÍCIO

PROPRIEDADES APÓS CHOQUE NORMAL

Uma onda de choque normal ocorre na seção de testes de um túnel de ventos supersônico. A montante da onda $M_1=3$, $P_1=0,5\text{atm}$ e $T_1=200\text{K}$. Encontre Mach, pressão, temperatura e massa específica após o choque.



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Uma onda de choque normal ocorre na seção de testes de um túnel de ventos supersônico. A montante da onda $M_1=3$, $P_1=0,5\text{atm}$ e $T_1=200\text{K}$. Encontre Mach, pressão, temperatura e massa específica após o choque.



$$\begin{array}{c}
 M_1 = 3 \\
 P_1 = 0,5 \text{ atm} = 50 \text{ kPa} \\
 T_1 = 200 \text{ K}
 \end{array}
 \quad
 \begin{array}{c}
 \text{---} \\
 | \\
 \text{---} \rightarrow
 \end{array}
 \quad
 M_2, T_2, P_2$$

$$\text{Ar} \left\{ \begin{array}{l} k = 1,4 \\ R = 287 \frac{\text{J}}{\text{kg K}} \end{array} \right.$$

$$P_1 = \rho_1 R T_1$$

$$\rho_1 = \frac{50 \cdot 10^3 \text{ Pa}}{287 \frac{\text{J}}{\text{kg K}} \cdot 200 \text{ K}}$$

$$\rho_1 = 0,87108 \text{ kg/m}^3$$

ρ_2

Formulário

Gráficos

Calculadora Online

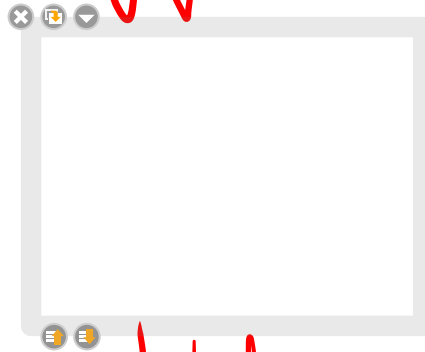
$$M_2^2 = \frac{1 + \left[\frac{k-1}{2}\right] M_1^2}{k M_1^2 - \frac{k-1}{2}}$$

$$\frac{\rho_2}{\rho_1} = \frac{V_1}{V_2} = \frac{(k+1) M_1^2}{2 + (k-1) M_1^2}$$

$$\frac{P_2}{P_1} = 1 + \frac{2k}{k+1} (M_1^2 - 1)$$

$$\frac{T_2}{T_1} = \frac{h_2}{h_1} = \left[1 + \frac{2k}{k+1} (M_1^2 - 1) \right] \left[\frac{2 + (k-1) M_1^2}{(k+1) M_1^2} \right]$$

$$s_2 - s_1 = -R \ln \frac{P_{o2}}{P_{o1}}$$



Tabelas

Tabela 1.2 – Tabela para Choque Normal. $k = 1,4$

M_1	M_2	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	ds	$\frac{P_{o2}}{P_{o1}}$	$\frac{P_{o2}}{P_{o1}}$
1,00	1,000000	1,000000	1,000000	1,000000	0,000000	1,000000	1,892929
1,02	0,980519	1,033441	1,047133	1,013249	0,000003	0,999990	1,937897
1,04	0,962025	1,067088	1,095200	1,026345	0,000022	0,999923	1,984423
1,06	0,944445	1,100921	1,144200	1,039312	0,000072	0,999751	2,032453
1,08	0,927713	1,134925	1,194133	1,052170	0,000163	0,999431	2,081942
1,10	0,911770	1,169082	1,245000	1,064938	0,000308	0,998928	2,132847
1,12	0,896563	1,203377	1,296800	1,077634	0,000513	0,998213	2,185134
1,14	0,882042	1,237793	1,349533	1,090274	0,000787	0,997261	2,238769
1,16	0,868162	1,272315	1,403200	1,102872	0,001135	0,996052	2,293725
1,18	0,854884	1,306927	1,457800	1,115441	0,001563	0,994569	2,349977
1,20	0,842170	1,341615	1,513333	1,127994	0,002074	0,992798	2,407502
1,22	0,829986	1,376364	1,569800	1,140541	0,002673	0,990731	2,466280
1,24	0,818301	1,411160	1,627200	1,153094	0,003361	0,988359	2,526294
1,26	0,807085	1,445989	1,685533	1,165661	0,004140	0,985677	2,587527
1,28	0,796312	1,480839	1,744800	1,178251	0,005014	0,982682	2,649964
1,30	0,785957	1,515695	1,805000	1,190873	0,005982	0,979374	2,713594

<http://www.dept.aoe.vt.edu/~devenpor/aoe3114/calc.html>

Formulação

MACH

Gráfico



$$M_2' = \frac{1 + \frac{k-1}{2} M_1'^2}{k M_1'^2 - \frac{(k-1)}{2}}$$

$$M_2' = \frac{1 + \frac{(1,4-1)}{2} \cdot 3'^2}{1,4 \cdot 3'^2 - \frac{(1,4-1)}{2}}$$

$$M_2 = 0,475$$

Fórmula

PRESSÃO

Gráfico



$$\frac{P_2}{P_1} = 1 + \frac{2k}{k+1} (M_1^2 - 1)$$

$$\frac{P_2}{P_1} = 1 + \frac{2 \cdot 1,4}{1,4 + 1} (3^2 - 1)$$

$$\frac{P_2}{P_1} = 10,33$$

$$P_2 = 5,165 \text{ atm}$$

Massa Específica

Formulação

Gráfico



$$\frac{\rho_2}{\rho_1} = \frac{(k+1) M_1^2}{2 + (k-1) M_1^2}$$

$$\frac{\rho_2}{\rho_1} = \frac{(1,4+1) \cdot 3^2}{2 + (1,4-1) \cdot 3^2}$$

$$\frac{\rho_2}{\rho_1} = 3,8571$$

$$\rho_2 = 3,3538 \frac{\text{kg}}{\text{m}^3}$$

TEMPERATURA

Formulação

$$p = \rho R T \leadsto T = \frac{p}{\rho R}$$

$$\frac{T_2}{p_2} = \frac{p_2}{\rho_2 R} \cdot \frac{\rho_1 R}{p_1} \leadsto \frac{T_2}{T_1} = \frac{p_2}{p_1} \cdot \frac{\rho_1}{\rho_2}$$

$$\frac{T_2}{T_1} = \left[1 + \frac{2k}{k+1} (M_1^2 - 1) \right] \left[\frac{2 + (k-1) M_1^2}{(k+1) M_1^2} \right]$$

$$\frac{T_2}{T_1} = 0,33 \cdot \frac{1}{3,8571}$$

$$\frac{T_2}{T_1} = 2,1190$$

$$T_2 = 535,8 \text{ K}$$

Gráfico

Propriedades de Estagnação e Críticas

$$M_1 = 3$$

$$p_1 = 0,5 \text{ atm}$$

$$T_1 = 200 \text{ K}$$

$$M_2 = 0,475$$

$$p_2 = 5,165 \text{ atm}$$

$$T_2 = 535,8 \text{ K}$$

$$T_{01}, p_{01}$$

$$T_1^*, p_1^*$$

$$T_{02}, p_{02}$$

$$T_2^*, p_2^*$$

$$\frac{T_0}{T} = \left(\frac{p_0}{p} \right)^{\frac{k-1}{k}} = \left(\frac{p_0}{p} \right)^{k-1}$$

$$\frac{T_0}{T^*} = \left(\frac{p_0}{p^*} \right)^{\frac{k-1}{k}} = \left(\frac{p_0}{p^*} \right)^{k-1}$$

Propriedades de Estagnação e Críticas

$$M_1 = 3$$

$$p_1 = 0,5 \text{ atm}$$

$$T_1 = 200 \text{ K}$$

$$M_2 = 0,475$$

$$p_2 = 5,165 \text{ atm}$$

$$T_2 = 535,8 \text{ K}$$

$$\frac{T_{01}}{T_1} = 1 + \frac{k-1}{2} M_1^2 \quad \leadsto \quad \frac{T_{01}}{T_1} = 2,8 \quad \leadsto \quad T_{01} = 560 \text{ K}$$

$$\frac{p_{01}}{p_1} = \left(\frac{T_{01}}{T_1} \right)^{k/k-1} \quad \rightarrow \quad \frac{p_{01}}{p_1} = 2,8^{3,5} = 36,13 \quad \leadsto \quad p_{01} = 18,36 \text{ atm}$$

$$\frac{\rho_{01}}{\rho_1} = \left(\frac{T_{01}}{T_1} \right)^{1/k-1} = 13,1188 \quad \rightarrow \quad \rho_{01} = 11,427 \text{ kg/m}^3$$

Propriedades de Estagnação e Críticas

$M_1 = 3$	$M_2 = 0,475$
$p_1 = 0,5 \text{ atm}$	$p_2 = 5,165 \text{ atm}$
$T_1 = 200 \text{ K}$	$T_2 = 535,8 \text{ K}$

$$\frac{T_{01}}{T_{1^*}} = \frac{k+1}{2} \Rightarrow T_{1^*} = 466,67 \text{ K}$$

$$\frac{p_{01}}{p_{1^*}} = \left(\frac{T_{01}}{T_{1^*}} \right)^{k/(k-1)} \Rightarrow p_{1^*} = 97 \text{ atm}$$

$$\frac{\rho_{01}}{\rho_{1^*}} = \left(\frac{T_{01}}{T_{1^*}} \right)^{1/(k-1)} \Rightarrow \frac{\rho_{01}}{\rho_{1^*}} = 1,57744 \Rightarrow \rho_{1^*} = 7,2440 \frac{\text{kg}}{\text{m}^3}$$

Propriedades de Estagnação e Críticas

$$\begin{aligned} M_1 &= 3 \\ p_1 &= 0,5 \text{ atm} \\ T_1 &= 200 \text{ K} \end{aligned}$$

$$\begin{aligned} M_2 &= 0,475 \\ p_2 &= 5,165 \text{ atm} \\ T_2 &= 535,8 \text{ K} \end{aligned} \quad \left. \vphantom{\begin{aligned} M_2 &= 0,475 \\ p_2 &= 5,165 \text{ atm} \\ T_2 &= 535,8 \text{ K} \end{aligned}} \right\} \rho_2 = 3,3588 \text{ kg/m}^3$$

$$\frac{T_{02}}{T_2} = 1 + \frac{k-1}{2} M_2^2 \rightarrow \frac{T_{02}}{T_2} = 1,045125 \rightarrow T_{02} = 560 \text{ K}$$

$$\frac{p_{02}}{p_2} = \left(\frac{T_{02}}{T_2} \right)^{k/(k-1)} \rightarrow \frac{p_{02}}{p_2} = 1,045125^{3,5} = 1,167 \rightarrow p_{02} = 6,03 \text{ atm}$$

$$\frac{\rho_{02}}{\rho_2} = \left(\frac{T_{02}}{T_2} \right)^{1/(k-1)} \rightarrow \frac{\rho_{02}}{\rho_2} = 1,1168 \rightarrow \rho_{02} = 3,751 \text{ kg/m}^3$$

Propriedades de Estagnação e Críticas

$$M_1 = 3$$

$$p_1 = 0,5 \text{ atm}$$

$$T_1 = 200 \text{ K}$$

$$M_2 = 0,475$$

$$p_2 = 5,165 \text{ atm}$$

$$T_2 = 535,8 \text{ K}$$

$$\rho_2 = 3,3588 \text{ kg/m}^3$$

$$\frac{T_2}{T_2^*} = \frac{k+1}{2} \Rightarrow T_2^* = 466,67 \text{ K} = T_1^*$$

$$\frac{p_2}{p_2^*} = \left(\frac{T_2}{T_2^*} \right)^{k/(k-1)} \Rightarrow p_2^* = 3,18 \text{ atm}$$

$$\frac{\rho_2}{\rho_2^*} = \left(\frac{T_2}{T_2^*} \right)^{1/(k-1)} \Rightarrow \frac{\rho_2}{\rho_2^*} = 1,57744 \Rightarrow \rho_2^* = 2,3773 \frac{\text{kg}}{\text{m}^3}$$

$$ds = c_p \ln\left(\frac{T_2}{T_1}\right) - R \ln\left(\frac{P_2}{P_1}\right)$$

ou

$$S_{20} - S_{10} = S_2 - S_1$$

$$\sim ds = c_p \ln\left(\frac{T_{20}}{T_{10}}\right) - R \ln\left(\frac{P_{02}}{P_{01}}\right) \leadsto ds = -R \ln\left(\frac{P_{02}}{P_{01}}\right)$$

$$+ ds = 1,0045 \frac{\text{kJ}}{\text{kg K}} \ln\left(\frac{535,8}{200}\right) - 0,287 \frac{\text{kJ}}{\text{kg K}} \ln\left(\frac{5,165}{0,5}\right) \leadsto ds = 0,3197 \frac{\text{kJ}}{\text{K}}$$

$$+ ds = -0,287 \frac{\text{kJ}}{\text{kg K}} \ln\left(\frac{6,03}{18,36}\right) \leadsto ds = 0,3196 \frac{\text{kJ}}{\text{K}}$$

Tabela 1.2 - Continuação da página anterior

M_1	M_2	$\frac{P_2}{P_1}$	$\frac{P_3}{P_1}$	$\frac{T_2}{T_1}$	ds	$\frac{P_{02}}{P_1}$	$\frac{P_{03}}{P_1}$
2,00	0,577350	2,666667	4,500000	1,687500	0,093933	0,720874	5,640441
2,02	0,573972	2,696181	4,593800	1,703817	0,097678	0,711527	5,743330
2,04	0,570679	2,725463	4,688533	1,720271	0,101473	0,702180	5,847268
2,06	0,567467	2,754511	4,784200	1,736860	0,105317	0,692839	5,952256
2,08	0,564334	2,783325	4,880800	1,753586	0,109207	0,683512	6,058292
2,10	0,561277	2,811902	4,978333	1,770450	0,113142	0,674203	6,165375
2,12	0,558294	2,840243	5,076800	1,787453	0,117122	0,664919	6,273505
2,14	0,555383	2,868345	5,176200	1,804594	0,121144	0,655666	6,382681
2,16	0,552541	2,896209	5,276533	1,821876	0,125208	0,646447	6,492903
2,18	0,549766	2,923834	5,377800	1,839297	0,129312	0,637269	6,604169
2,20	0,547056	2,951220	5,480000	1,856860	0,133454	0,628136	6,716480
2,22	0,544409	2,978365	5,583133	1,874563	0,137635	0,619053	6,829835
2,24	0,541822	2,995271	5,687200	1,892401	0,141852	0,610023	6,944232
2,26	0,539295	3,011936	5,791200	1,910373	0,146105	0,601051	7,059673
2,28	0,536825	3,028342	5,895133	1,928527	0,150391	0,592140	7,176155
2,30	0,534411	3,084548	6,005000	1,946801	0,154711	0,583295	7,293680
2,32	0,532051	3,110495	6,112800	1,965218	0,159063	0,574517	7,412245
2,34	0,529743	3,136202	6,221533	1,983779	0,163446	0,565810	7,531852
2,36	0,527486	3,161671	6,331200	2,002485	0,167858	0,557177	7,652499
2,38	0,525278	3,186942	6,441800	2,021336	0,172303	0,548611	7,774187
2,40	0,523118	3,212046	6,553333	2,040332	0,176777	0,540104	7,896914
2,42	0,521004	3,236853	6,665800	2,059473	0,181265	0,531748	8,020681
2,44	0,518936	3,261174	6,779200	2,078760	0,185787	0,523435	8,145487
2,46	0,516911	3,285461	6,893533	2,098194	0,190334	0,515208	8,271331
2,48	0,514929	3,309514	7,008800	2,117773	0,194905	0,507067	8,398214
2,50	0,512989	3,333333	7,125000	2,137500	0,199499	0,499015	8,526136
2,52	0,511089	3,356921	7,242133	2,157374	0,204116	0,491052	8,655095
2,54	0,509228	3,380279	7,360200	2,177394	0,208754	0,483181	8,785092
2,56	0,507406	3,403407	7,479200	2,197563	0,213412	0,475402	8,916127
2,58	0,505620	3,426307	7,599133	2,217879	0,218090	0,467715	9,048199
2,60	0,503871	3,448980	7,720000	2,238343	0,222787	0,460123	9,181308
2,62	0,502157	3,471427	7,841800	2,258956	0,227502	0,452625	9,315453
2,64	0,500477	3,493651	7,964533	2,279717	0,232235	0,445223	9,450636
2,66	0,498830	3,515651	8,088200	2,300626	0,236984	0,437916	9,586854
2,68	0,497216	3,537431	8,212800	2,321685	0,241749	0,430705	9,724109
2,70	0,495634	3,558991	8,338333	2,342892	0,246530	0,423590	9,862399
2,72	0,494082	3,580333	8,464800	2,364249	0,251325	0,416572	10,001726
2,74	0,492560	3,601458	8,592200	2,385756	0,256133	0,409650	10,142088
2,76	0,491068	3,622369	8,720533	2,407412	0,260955	0,402825	10,283485
2,78	0,489604	3,643066	8,849800	2,429218	0,265790	0,396096	10,425918
2,80	0,488167	3,663551	8,980000	2,451173	0,270636	0,389464	10,569386
2,82	0,486758	3,683827	9,111133	2,473279	0,275494	0,382927	10,713888
2,84	0,485376	3,703894	9,243200	2,495536	0,280363	0,376486	10,859426
2,86	0,484019	3,723755	9,376200	2,517942	0,285241	0,370141	11,005998
2,88	0,482687	3,743411	9,510133	2,540500	0,290130	0,363890	11,153605
2,90	0,481380	3,762864	9,645000	2,563207	0,295027	0,357733	11,302246
2,92	0,480096	3,782115	9,780800	2,586066	0,299933	0,351670	11,451922
2,94	0,478836	3,801167	9,917533	2,609076	0,304846	0,345701	11,602632
2,96	0,477599	3,820021	10,055200	2,632237	0,309767	0,339823	11,754375
2,98	0,476384	3,838679	10,193800	2,655549	0,314696	0,334038	11,907153
3,00	0,475191	3,857143	10,333333	2,679012	0,319630	0,328344	12,060965

$$M_1 = 3$$

$$M_2 = 0,475191$$

$$\frac{P_2}{P_1} = 3,857143$$

$$\frac{P_2}{P_1} = 10,333$$

$$\frac{T_2}{T_1} = 2,679012$$

$$ds = 0,31963$$

$$\frac{P_{02}}{P_1} = 12,060965$$

$$\text{Equação}$$

$$(0,4751)$$

$$(3,8571)$$

$$(10,33)$$

$$(2,6790)$$

$$(0,3196)$$

$$\left(\frac{6,03}{0,5} = 12,06 \right)$$

Propriedades de Estagnação 1

$$\begin{array}{l|l} M_1 = 3 & M_2 = 0,475 \\ P_1 = 0,5 \text{ atm} & P_2 = 5,165 \text{ atm} \\ T_1 = 200 \text{ K} & T_2 = 535,8 \text{ K} \end{array}$$

Tabela 1.1 – Tabela para escoamento Isoentrópico. $k = 1,4$

M	$\frac{T_0}{T}$	$\frac{P_0}{P}$	$\frac{\rho_0}{\rho}$
2,02	1,816080000	8,071837587	4,444648686
2,04	1,832320000	8,327307979	4,544679957
2,06	1,848720000	8,591104239	4,647055389
2,08	1,865280000	8,863476911	4,751821127
2,10	1,882000000	9,144683243	4,859024040
2,12	1,898880000	9,434987325	4,968711727
2,14	1,915920000	9,734660242	5,080932524
2,90	2,682000000	31,594079549	11,780044575
2,92	2,705280000	32,564377179	12,037340748
2,94	2,728720000	33,562662760	12,299782594
2,96	2,752320000	34,589655845	12,567454309
2,98	2,776080000	35,646091744	12,840441106
3,00	2,800000000	36,732721805	13,118829216
3,02	2,824080000	37,850313686	13,402705903

Continua na página posterior

$$\frac{T_0}{T_1} = 2,8 \quad (2,8)$$

$$\frac{P_0}{P_1} = 36,732721805 \quad (36,73)$$

$$\frac{\rho_0}{\rho_1} = 13,1188 \quad (13,1188)$$

Propriedades Críticas ↓

$$\begin{array}{l|l} M_1 = 3 & M_2 = 0,475 \\ p_1 = 0,5 \text{ atm} & p_2 = 5,165 \text{ atm} \\ T_1 = 200 \text{ K} & T_2 = 535,8 \text{ K} \end{array}$$

Tabela 1.1 – Tabela para escoamento Isoentrópico. $k = 1,4$

M	$\frac{T_0}{T}$	$\frac{P_0}{P}$	$\frac{\rho_0}{\rho}$
1,00	1,200000000	1,892929159	1,577440966
1,02	1,208080000	1,937915922	1,604128801
1,04	1,216320000	1,984574779	1,631622253
1,06	1,224720000	2,032960007	1,659938604
1,08	1,233280000	2,083127778	1,689095565
1,10	1,242000000	2,135136208	1,719111279

$$\frac{T_0}{T_1^*} = 1,2$$

$$\frac{T_0}{T^*} = \frac{k+1}{2} = 1,2$$

$$\frac{P_0}{P_1^*} = 1,892929159$$

$$\left(\frac{18,36}{9,7} = 1,894783 \right)$$

$$\frac{\rho_0}{\rho_1^*} = 1,577440966 \checkmark$$

Propriedades de Estagnação 2

$$\begin{array}{l|l} M_1 = 3 & M_2 = 0,475 \\ P_1 = 0,5 \text{ atm} & P_2 = 5,165 \text{ atm} \\ T_1 = 200 \text{ K} & T_2 = 535,8 \text{ K} \end{array}$$

Tabela 1.1 – Tabela para escoamento Isoentrópico. $k = 1,4$

M	$\frac{T_0}{T}$	$\frac{P_0}{P}$	$\frac{\rho_0}{\rho}$
0,00	1,000000000	1,000000000	1,000000000
0,02	1,000080000	1,000280028	1,000200012
0,04	1,000320000	1,001120448	1,000800192
0,06	1,000720000	1,002522269	1,001800972
0,08	1,001280000	1,004487173	1,003203073
0,10	1,002000000	1,007017518	1,005007502
0,12	1,002880000	1,010116340	1,007215559
0,14	1,003920000	1,013787360	1,009828831
0,16	1,005120000	1,018034982	1,012849194
0,18	1,006480000	1,022864304	1,016278817
0,20	1,008000000	1,028281121	1,020120160
0,22	1,009680000	1,034291935	1,024375975
0,24	1,011520000	1,040903957	1,029049309
0,26	1,013520000	1,048125123	1,034143503
0,28	1,015680000	1,055964098	1,039662194
0,30	1,018000000	1,064430286	1,045609318
0,32	1,020480000	1,073533846	1,051989110
0,34	1,023120000	1,083285700	1,058806103
0,36	1,025920000	1,093697545	1,066065136
0,38	1,028880000	1,104781869	1,073771352
0,40	1,032000000	1,116551966	1,081930199
0,42	1,035280000	1,129021948	1,090547435
0,44	1,038720000	1,142206766	1,099629126
0,46	1,042320000	1,156122222	1,109181654
0,48	1,046080000	1,170784991	1,119211715

$$\begin{array}{l} M \\ 0,46 \\ 0,475 \\ 0,48 \end{array} \quad \begin{array}{l} 1,04232 \\ T_0/T_2 \\ 1,04608 \end{array} \quad \begin{array}{l} 1,156122 \\ P_0/P_2 \\ 1,170784991 \end{array} \quad \begin{array}{l} 1,109181654 \\ g_0/g_2 \\ 1,119211715 \end{array}$$

$$\begin{array}{l} \frac{T_0}{T_2} = 1,0442 \quad (1,045125) \\ \frac{P_0}{P_2} = 1,1634 \quad (1,167) \\ \frac{g_0}{g_2} = 1,1142 \quad (1,1167) \end{array}$$

Propriedades Críticas 2

$$\begin{array}{l|l} M_1 = 3 & M_2 = 0,475 \\ p_1 = 0,5 \text{ atm} & p_2 = 5,165 \text{ atm} \\ T_1 = 200 \text{ K} & T_2 = 535,8 \text{ K} \end{array}$$

Tabela 1.1 – Tabela para escoamento Isoentrópico. $k = 1,4$

M	$\frac{T_0}{T}$	$\frac{p_0}{p}$	$\frac{\rho_0}{\rho}$
1,00	1,200000000	1,892929159	1,577440966
1,02	1,208080000	1,937915922	1,604128801
1,04	1,216320000	1,984574779	1,631622253
1,06	1,224720000	2,032960007	1,659938604
1,08	1,233280000	2,083127778	1,689095565
1,10	1,242000000	2,135136208	1,719111279

$$\frac{T_0}{T_2} = 1,2 \quad \left(\frac{560}{466,67} = 1,2 \right)$$

$$\frac{p_0}{p_2} = 1,892929159 \quad \left(\frac{6,03}{3,18} = 1,896226 \right)$$

CALCULADORA ONLINE $M_1 = 3$

$$M_2 = 0,475191 \quad (0,475)$$

$$\frac{g_2}{g_1} = 3,857143 \quad (3,8571)$$

$$\frac{p_2}{p_1} = 10,33\bar{3} \quad (10,3\bar{3})$$

$$\frac{T_2}{T_1} = 2,679012 \quad (2,6790)$$

$$ds = 0,31963 \quad (0,3196)$$

$$\frac{p_{02}}{p_1} = 12,060965 \quad \left(\frac{6,03}{0,5} = 12,06 \right)$$

Propriedades de Estagnação e Críticas I

$$M_1 = 3$$

$$p_1 = 0,5 \text{ atm}$$

$$T_1 = 200 \text{ K}$$

$$M_2 = 0,475$$

$$p_2 = 5,165 \text{ atm}$$

$$T_2 = 535,8 \text{ K}$$

$$\frac{T_{01}}{T_1} = 2,8 \leadsto \frac{T_1}{T_{01}} = 0,35714$$

$$\frac{p_{01}}{p_1} = 36,73 \leadsto \frac{p_1}{p_{01}} = 0,02722$$

$$\frac{p_1}{p_1^*} = 0,0515325 \leadsto p_1^* = 3,70 \text{ atm}$$

$$\frac{T_1}{T_1^*} = 0,42851142 \leadsto T_1^* = 466,61 \text{ K}$$

Propriedades de Estagnação e Críticas 2

$$M_1 = 3$$

$$p_1 = 0,5 \text{ atm}$$

$$T_1 = 200 \text{ K}$$

$$M_2 = 0,475$$

$$p_2 = 5,365 \text{ atm}$$

$$T_2 = 535,8 \text{ K}$$

Compressible Aerodynamics **Calculator** 3.0

[What's this?](#) [Smartphone Version \(1/31/11\)](#)... [HP Prime version \(8/25/14\)](#)... [Recently added](#) [Other Java](#)

Isentropic Flow Relations Perfect Gas, Gamma = , angles in degrees.

INPUT: Mach number =

Mach number=	0.475	Mach angle=		P-M angle=	
p/p ₀ =	0.85686257	rho/rho ₀ =	0.89552849	T/T ₀ =	0.95682334
p/p*=	1.62198014	rho/rho*=	1.41264333	T/T*=	1.14818801
A/A*=					1.39080858

Normal Shock Relations Perfect Gas, Gamma =

INPUT: M1 =

M ₁ =		M ₂ =		p ₀₂ /p ₀₁ =		p ₁ /p ₀₂ =	
p ₂ /p ₁ =		rho ₂ /rho ₁ =		T ₂ /T ₁ =			

$$\frac{T_{02}}{T_2} = 1,045115 \leadsto \frac{T_2}{T_{02}} = 0,9568233$$

$$\frac{p_{02}}{p_2} = 1,167 \leadsto \frac{p_2}{p_{02}} = 0,856838...$$

$$M_1 = 3$$

$$P_1 = 0,5 \text{ atm}$$

$$T_1 = 200 \text{ K}$$

$$T_{01} = 560 \text{ K}$$

$$P_{01} = 18,36 \text{ atm}$$

$$\rho_{01} = 11,427 \text{ kg/m}^3$$

$$T_1^* = 466,67 \text{ K}$$

$$P_1^* = 37 \text{ atm}$$

$$\rho_1^* = 7,2440 \text{ kg/m}^3$$

$$M_2 = 0,415$$

$$P_2 = 5,165 \text{ atm}$$

$$T_2 = 535,8 \text{ K}$$

$$T_{02} = 560 \text{ K}$$

$$P_{02} = 4,03 \text{ atm}$$

$$\rho_{02} = 3,751 \text{ kg/m}^3$$

$$T_2^* = 466,67 \text{ K}$$

$$P_2^* = 3,18 \text{ atm}$$

$$\rho_2^* = 2,3779 \text{ kg/m}^3$$

The logo consists of a large yellow circle centered on a red background. Inside the circle, the text "CIÊNCIAS TÉRMICAS" is written in a bold, white, sans-serif font, with "CIÊNCIAS" on the top line and "TÉRMICAS" on the bottom line. Below this, the text ".com" is written in a smaller, white, sans-serif font.

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